

A Vector Approach for Noise Parameter Fitting and Selection of Source Admittances

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Abstract—Simple vector concepts can be used to determine noise parameters from measured data. The use of such concepts leads to a simplification in the least-square fitting algorithm, complete determination of the admittance loci that produce ill conditioning, and simple criteria for the selection of source admittances that minimize the sensitivity of the noise parameters to experimental error. The sensitivity of the noise parameters to small perturbations in the reflection coefficients is compared for a group of source admittances selected with the techniques described here and a group of admittances presented in a previous work. The results show that a great reduction in the error of the noise parameters can be achieved by properly selecting the source admittances.

I. INTRODUCTION

THE dependence of the noise factor of a two-port on the source admittance is given by

$$F = F_{\text{MIN}} + \frac{R_N}{G_S} \left[(G_S - G_0)^2 + (B_S - B_0)^2 \right] \quad (1)$$

where

F_{MIN} = minimum noise figure of the device,
 $G_0 + jB_0$ = source admittance for minimum noise figure,
 R_N = noise resistance,
 $G_S + jB_S$ = source admittance.

Using this equation, the noise parameters (F_{MIN} , R_N , G_0 , B_0) can be determined if F is measured with at least four different sets of source admittances ($G_S + jB_S$) [1].

Most of the algorithms already developed use more than four data sets to minimize the effect of measurement errors. However, depending on the selection of the source admittances, these algorithms may produce ill conditioning, i.e., a strong dependence of the results (noise parameters) on small perturbations in the data (source reflection coefficients, measured noise figures) caused by measurement and/or computation errors, and inaccurate extraction of parasitic elements. Ill conditioning occurs whenever the source admittances lie very close to one of the loci derived in Appendix II.

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Computer simulations in a previous work [2] indicated that a proper distribution of source impedances in the Smith chart (rather than an increase in the number of impedances) is necessary to minimize the fitting errors; however, no selection criteria were given.

The algorithm presented in this work uses a vector approach which leads to a quasi-graphical interpretation of the fitting process and an improved understanding of the ill-conditioning phenomenon. A quantitative description of the degree of ill conditioning that a group of source admittances produces comes naturally from this formulation, as well as criteria for the selection of source admittances that avoid ill conditioning. The proposed formulation accepts redundant and nonredundant data and least-squares fitting is performed without an iterative search.

II. FORMULATION: PROJECTION THEOREM

Equation (1) can be easily rearranged to

$$F = (F_{\text{MIN}} - 2R_N G_0) + R_N \frac{G_S^2 + B_S^2}{G_S} - 2R_N B_0 \left(\frac{B_S}{G_S} \right) + [R_N (G_0^2 + B_0^2)] \frac{1}{G_S} \quad (2)$$

where each of the four terms has a different dependence on the source admittance. At this point, it is convenient to define the following vectors:

$$\begin{aligned} i &= 1 \cdots n \\ \bar{F}_M &= (F_{M1}, F_{M2}, \dots, F_{Mi}, \dots, F_{Mn})^T \\ \bar{V}_1 &= (1, 1 \cdots 1)^T \\ \bar{V}_2 &= \left(\frac{G_{s1}^2 + B_{s1}^2}{G_{s1}}, \frac{G_{s2}^2 + B_{s2}^2}{G_{s2}}, \dots, \right. \\ &\quad \left. \frac{G_{sn}^2 + B_{sn}^2}{G_{sn}}, \dots, \frac{G_{sn}^2 + B_{sn}^2}{G_{sn}} \right) \\ \bar{V}_3 &= \left(\frac{B_{s1}}{G_{s1}}, \frac{B_{s2}}{G_{s2}}, \dots, \frac{B_{si}}{G_{si}}, \dots, \frac{B_{sn}}{G_{sn}} \right) \\ \bar{V}_4 &= \left(\frac{1}{G_{s1}}, \frac{1}{G_{s2}}, \dots, \frac{1}{G_{si}}, \dots, \frac{1}{G_{sn}} \right) \quad (3) \end{aligned}$$

where F_{Mi} , G_{Si} , and B_{Si} are the measured noise figure,

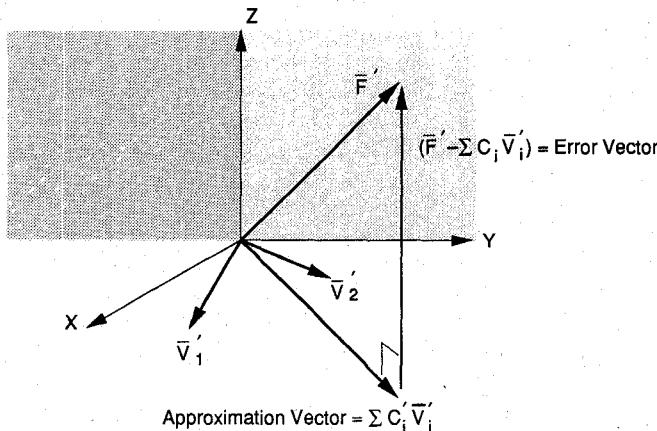


Fig. 1. Projection theorem in a 3-D space with two approximating vectors \bar{V}_1' , \bar{V}_2' . The magnitude of the error vector is minimum when such vector is perpendicular to the plane defined by \bar{V}_1' and \bar{V}_2' .

conductance, and susceptance, respectively, associated with the i th measurement set, and n is the total number of sets measured. A proper fitting of (2) for all the n different data sets is equivalent to the following vector approximation:

$$\bar{F}_M \approx C_1 \bar{V}_1 + C_2 \bar{V}_2 + C_3 \bar{V}_3 + C_4 \bar{V}_4 \quad (4)$$

where

$$\begin{aligned} C_1 &= (F_{\text{MIN}} - 2R_N G_0) & C_2 &= R_N \\ C_3 &= -2R_N B_0 & C_4 &= R_N (G_0^2 + B_0^2). \end{aligned} \quad (5)$$

The problem of finding the noise parameters for a best fit of (1) can now be reduced to finding the coefficients $C_1 \dots C_4$ to minimize the error vector between F_M and the linear combination of vectors V_i . The magnitude of this error vector is given by

$$\epsilon = \left[\sum_{i=1}^n (F_{Mi} - F_{\text{fitted } i})^2 \right]^{1/2} \quad (6)$$

where

$$F_{\text{fitted } i} = \sum_{j=1}^4 C_j V_{ji}, \quad i = 1, 2, \dots, n$$

and V_{ji} is the i th component of \bar{V}_j . This problem can be solved with the help of the projection theorem in Hilbert spaces (see Fig. 1), which states that the magnitude of the error vector is minimum when such a vector is orthogonal to all vectors \bar{V}_i , i.e.,

$$\left\langle \left(F_M - \sum_{j=1}^4 C_j V_j \right), V_i \right\rangle = 0, \quad i = 1, \dots, 4 \quad (7)$$

where $\langle \cdot \rangle$ indicates the inner product of two vectors. By using basic properties of the inner product, (7) can be rearranged to

$$\sum_{j=1}^4 \langle \bar{V}_i, \bar{V}_j \rangle C_j = \langle \bar{F}_M, \bar{V}_i \rangle, \quad i = 1, \dots, 4. \quad (8)$$

Equation (8) defines a system of linear equations from which the values of the C_j 's can be found using standard techniques. However, advantage can be taken of the fact that the coefficient matrix in this system is symmetric and positive definite. This allows the use of Cholesky's method [3] for solving a linear system of equations, which leads to closed-form equations for the coefficients C_j (refer to Appendix I).

Once these coefficients are found, the noise parameters can be determined through

$$\begin{aligned} R_N &= C_2 \\ B_0 &= -C_3 / 2R_N \\ G_0 &= \sqrt{\frac{C_4}{R_N} - B_0^2} \end{aligned}$$

and

$$F_{\text{MIN}} = C_1 + 2R_N G_0. \quad (9)$$

The results obtained up to this point are very similar to those reported in [4], where the vector formulation is not used. For least-squares fitting purposes, such formulation has the advantage of providing a simple description of the linear system that allows the use of Cholesky's formulas for the coefficients C_j . However, the major advantage of the vector approach can be found in the prevention of ill conditioning and minimization of the sensitivity of the noise parameters to measurement errors.

III. ILL CONDITIONING AND ERROR SENSITIVITY

Ill conditioning occurs when the vectors $\bar{V}_1 \dots \bar{V}_4$ in (3) are not linearly independent. Then the coefficients $C_1 \dots C_4$ in (4) are not uniquely defined. The 3-D equivalent to this case (Fig. 1) occurs when the vectors \bar{V}_1' and \bar{V}_2' are aligned. The orthogonal projection of \bar{F}' onto this line is still uniquely defined, but the coefficients C_1 and C_2 , which relate this projection to a linear combination of \bar{V}_1' and \bar{V}_2' , are not.

There are 11 possible ways in which the vectors $\bar{V}_1 \dots \bar{V}_4$ in (3) may not be linearly independent:

$$\begin{aligned} \bar{V}_1 &= \alpha_1 \bar{V}_2 & \bar{V}_1 &= \alpha_2 \bar{V}_3 \\ \bar{V}_1 &= \alpha_3 \bar{V}_4 & \bar{V}_2 &= \alpha_4 \bar{V}_3 \\ \bar{V}_2 &= \alpha_5 \bar{V}_4 & \bar{V}_3 &= \alpha_6 \bar{V}_4 \\ \bar{V}_1 &= \alpha_7 \bar{V}_2 + \beta_7 \bar{V}_3 & \\ \bar{V}_1 &= \alpha_8 \bar{V}_2 + \beta_8 \bar{V}_4 & \\ \bar{V}_1 &= \alpha_9 \bar{V}_3 + \beta_9 \bar{V}_4 & \\ \bar{V}_2 &= \alpha_{10} \bar{V}_3 + \beta_{10} \bar{V}_4 & \\ \bar{V}_1 &= \alpha_{11} \bar{V}_2 + \beta_{11} \bar{V}_3 + \gamma_{11} \bar{V}_4 \end{aligned} \quad (10)$$

where α_i , β_i , and γ_i are constants (other than zero) with proper dimensions. Each of the vector equations in (10) results in a family of admittance loci that can be represented as lines or circles in the admittance plane (refer to

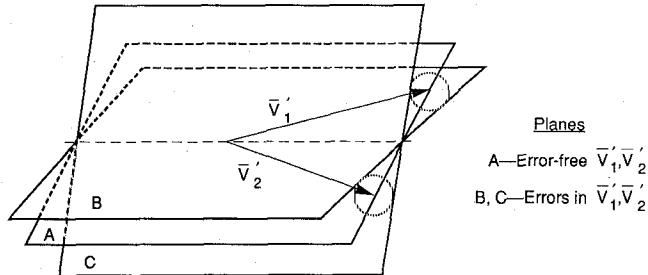


Fig. 2. Effect of the errors on the plane defined by \bar{V}_1' and \bar{V}_2' . Error sensitivity is minimized when \bar{V}_1' and \bar{V}_2' are orthogonal.

Appendix II). The determination of the noise parameters will be ill conditioned whenever all the admittances lie on—or are very close to—one of these loci.

Caruso and Sannino [5] describe a technique to avoid ill conditioning based on distributing the source admittances along two different ill-conditioning loci belonging to the same family. This is equivalent to avoiding only one condition in (10). To ensure that no ill conditioning will occur, none of the 11 conditions in (10) should be satisfied, or, equivalently, no ill conditioning locus will come close to all selected source admittances. Since there are 11 different families of ill-conditioning loci and they have up to three degrees of freedom, this is difficult to check.

The vector approach presented here allows simple selection criteria for the source admittances that guarantee the linear independence among all vectors $\bar{V}_1 \cdots \bar{V}_4$ and therefore ensures that none of the equations in (10) will hold. These criteria also minimize the sensitivity of the results (noise parameters) to perturbations in the data (source reflection coefficients and noise figures). Further insight into the mechanisms that translate errors from the data to the results is needed to establish these selection criteria.

In the 3-D analogy of Fig. 1, errors in the components of the vectors \bar{V}_1' and \bar{V}_2' generate an uncertainty in the plane that they define. This causes an uncertainty in the projection of the vector \bar{F}' onto this plane, which is the ultimate cause of errors in the coefficients C'_1 and C'_2 . If there is some degree of freedom in the selection of V_1' and V_2' (as in V_2 , V_3 , and V_4 in (3)), they should be chosen so that errors in their components have a minimum effect in the plane that they define.

Fig. 2 illustrates the changes in a plane caused by perturbations in the two vectors that define it. When no errors are present in V_1' and V_2' , they generate the plane A , whereas if some uncertainty exists (represented by dotted spheres in Fig. 2) the resulting plane could have any orientation included between planes B and C . The angle between these two planes is a measure of the uncertainty in plane A caused by uncertainties in V_1' and V_2' . This angle increases when the uncertainty in V_1' and V_2' (radii of the dotted spheres in Fig. 2) increases or when \bar{V}_1' and \bar{V}_2' tend to be aligned. For a given uncertainty in \bar{V}_1' and \bar{V}_2' , minimum uncertainty in the plane that they define is obtained when the two vectors are

orthogonal; for a given angle between \bar{V}_1' and \bar{V}_2' , the uncertainty in the plane can be reduced by making the ratio of the vector magnitude to the error magnitude large.

The above discussion can be applied to the selection of the source admittances that define \bar{V}_2 , \bar{V}_3 , and \bar{V}_4 in (3) as follows:

- 1) The uncertainty in \bar{V}_2 , \bar{V}_3 , and \bar{V}_4 should not be heavily dependent on the uncertainty in the values of the source reflection coefficient. Moreover, the uncertainty in any vector should be much smaller than its magnitude.
- 2) Ideally, $\bar{V}_1 \cdots \bar{V}_4$ should be orthogonal. In practice, the components of these vectors are dictated by the value of the source admittances (eq. (3)) and they cannot be chosen to achieve complete orthogonality. The degree of orthogonality between two vectors is given by

$$\cos(\bar{V}_i, \bar{V}_j) = \frac{\langle \bar{V}_i, \bar{V}_j \rangle}{|\bar{V}_i| \cdot |\bar{V}_j|} \quad (i \neq j). \quad (11)$$

Equation (11) can be identified as the cosine of the angle between \bar{V}_i and \bar{V}_j in a 3-D space. In our case there are six possible combinations of $\cos(\bar{V}_i, \bar{V}_j)$. Selection of the source admittances should involve a simultaneous minimization of the magnitude of all these factors. Note that this is equivalent to maximizing the diagonal terms of the coefficient matrix in (8). Linear system theory [3] shows that this will prevent a strong dependence of the resulting noise parameters on errors in both source admittances and measured noise figures.

IV. EXAMPLE: PRACTICAL SELECTION CRITERIA

In this section, a selection procedure is described to choose a set of seven source admittances. As shown in (3), the components of \bar{V}_1 , \bar{V}_2 , and \bar{V}_4 are always positive. This prevents their dot products from being zero; however, good orthogonality among these vectors can be obtained if \bar{V}_2 and \bar{V}_4 have one component much larger than the rest and the dominant component in \bar{V}_2 does not correspond to the one in \bar{V}_4 . In this case, the admittances are chosen to satisfy

$$\frac{G_{S1}^2 + B_{S1}^2}{G_{S1}} \gg \frac{G_{Si}^2 + B_{Si}^2}{G_{Si}}, \quad i \neq 1$$

$$\frac{1}{G_{S2}} \gg \frac{1}{G_{Si}}, \quad i \neq 2 \quad (12)$$

under these assumptions:

$$\cos(\bar{V}_1, \bar{V}_2) \cong \cos(\bar{V}_1, \bar{V}_4) \cong \frac{1}{|\bar{V}_1|} = \frac{1}{\sqrt{n}}$$

$$\cos(\bar{V}_2, \bar{V}_4) \cong \frac{[1 + (B_{S1}/G_{S1})^2][1 + (B_{S2}/G_{S2})^2]}{\left(\frac{G_{S1}^2 + B_{S1}^2}{G_{S1}}\right) \cdot \left(\frac{1}{G_{S2}}\right)} \quad (13)$$

B_{S1} and B_{S2} are chosen to be zero to minimize $\cos(\bar{V}_2, \bar{V}_4)$. The remaining source admittances have to be chosen to satisfy (12) and to ensure good orthogonality of \bar{V}_3 with \bar{V}_1 , \bar{V}_2 , and \bar{V}_4 . A possible choice is to make $B_{S3} = 0$ and $G_{S3} = 20$ mS (matched load) and $B_{Si} = -B_{S(i+1)}$ and $G_{Si} = G_{S(i+1)}$, $i = 4, 6, \dots$ (conjugate loads). The final group of source reflection coefficients to be presented to the transistor has been determined to be

$$\begin{aligned} \Gamma_{S1} &= 0.7 < 180^\circ & \Gamma_{S2} &= 0.7 < 0 & \Gamma_{S3} &= 0 \\ \Gamma_{S4} &= 0.3 < -90^\circ & \Gamma_{S5} &= 0.3 < +90^\circ \\ \Gamma_{S6} &= 0.6 < -90^\circ & \Gamma_{S7} &= 0.6 < +90^\circ. \end{aligned} \quad (14)$$

Gradient optimization has been used to obtain this result. The function minimized was defined as the maximum absolute value of six possible $\cos(\bar{V}_i, \bar{V}_j)$ factors. For the values shown in (14), this maximum was determined by $\cos(\bar{V}_1, \bar{V}_2) = \cos(\bar{V}_1, \bar{V}_4) = 0.76$ (equivalent to 40.5° in a 3-D space). This figure could be further reduced by allowing $|\Gamma_{S1}|$ and $|\Gamma_{S2}|$ to take a higher value. This would increase the first component in \bar{V}_2 and the second component in \bar{V}_4 , improving the orthogonality of both vectors with \bar{V}_1 ; however, small errors in Γ_{S1} when its magnitude is large provide large variations in G_{S1} and B_{S1} and, therefore, a high uncertainty in \bar{V}_2 , whose first component is dominant with respect to the others. Similarly, small variations in Γ_{S2} when its magnitude is large provide large variations in $1/G_{S2}$ and high uncertainty in \bar{V}_4 ; therefore, it is important not to increase Γ_{S1} and Γ_{S2} beyond the value for which acceptable orthogonality is achieved.

The values of Γ_{S4} and Γ_{S5} were optimized simultaneously in order to maintain their complementary nature. This was also done for Γ_{S6} and Γ_{S7} . In both cases, optimum orthogonality was obtained when the magnitude of the reflection coefficient was zero or unrealistically small. This solution was not acceptable since the magnitude of \bar{V}_3 was too small, making the uncertainty in this vector comparable to its magnitude. Minimum boundaries for $|\Gamma|$ had to be set in the optimization of Γ_{S4} , Γ_{S5} and Γ_{S6} , Γ_{S7} to avoid this effect ($|\Gamma_{S4}|_{\text{MIN}} = |\Gamma_{S5}|_{\text{MIN}} = 0.3$, $|\Gamma_{S6}|_{\text{MIN}} = |\Gamma_{S7}|_{\text{MIN}} = 0.6$). The optimum phase was found to be $\pm 90^\circ$ regardless of the limits in the magnitude. For these phases, $\cos(\bar{V}_1, \bar{V}_2) = \cos(\bar{V}_1, \bar{V}_4)$.

The set of source admittances in (14) was intended to be used in an experimental determination of our $0.25 \mu\text{m}$ MESFET devices at Ka band. Some of these source admittances had to be reselected because either the device was not stable or its measured noise figure was judged too high to allow an accurate reading of the noise figure meter; however, the criteria given in this and the previous section proved to be helpful in ruling out admittances that would have given rise to ill conditioning.

V. SOFTWARE DEVELOPED FOR NOISE PARAMETER FITTING

The essential features of the software developed for noise parameter fitting are shown in Fig. 3. First, the six factors $\cos(\bar{V}_i, \bar{V}_j)$ are calculated at each frequency point.

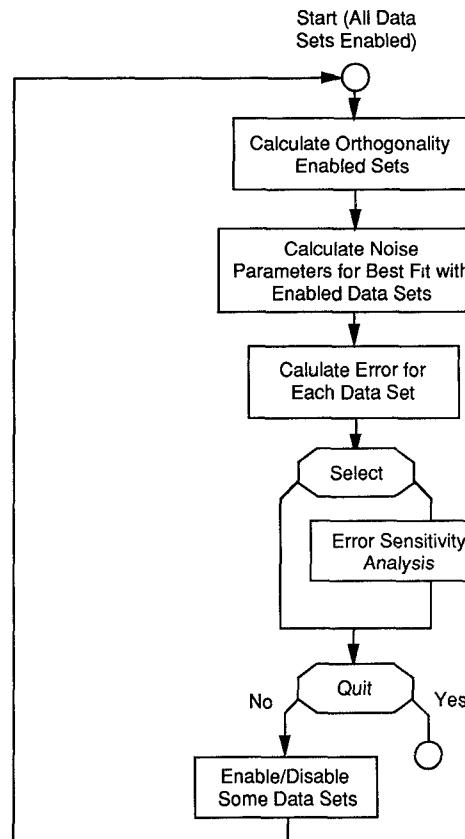


Fig. 3. Block diagram of the software developed for noise parameter fitting.

TABLE I
COMPARISON OF RESULTS WITH DIFFERENT CHOICES OF ENABLED DATA SETS

Enabled Data Sets	F_{min} (dB)	R_N (Ω)	$1/G_0$ (Ω)	$1/B_0$ (Ω)	$\text{ERR} \times 10^4$
3, 4, 7, 9	0.4567	6.077	38.20	40.65	8.8
1, 4, 5, 6	0.4539	4.893	41.71	39.65	3.8
All	0.4576	4.654	41.05	39.56	3.3

This gives the user an indication of how ill conditioned the system is at each frequency. Second, the noise parameters are calculated at each frequency using (8) and (9). Once the tentative noise parameters are known, they are used to calculate the tentative noise figure (F_{calc}) for each measurement set with (1). When F_{calc} is known, an error function is calculated for each set using

$$\text{ERR} = \frac{1}{n_f} \sqrt{\sum_{n_f} \left(\frac{F_{\text{meas}} - F_{\text{calc}}}{F_{\text{meas}}} \right)^2} \quad (15)$$

n_f being the number of frequency points.

A table is then generated displaying the values of these error functions. Based on this information, the user has the option of disabling some of the data sets and restarting the fitting process. This provides protection against errors not evenly distributed among data sets and an opportunity to alter the values of $\cos(V_i, V_j)$.

TABLE II
ORTHOGONALITY AND ERROR SENSITIVITY FOR TWO DIFFERENT GROUPS OF SOURCE ADMITTANCES

Case	$\cos(V_1, V_2)$	$\cos(V_1, V_3)$	$\cos(V_1, V_4)$	$\cos(V_2, V_3)$	$\cos(V_2, V_4)$	$\cos(V_3, V_4)$
1	0.97913	0.98398	0.99155	0.99193	0.95534	0.97627
2	0.75964	0	0.75964	0	0.33081	0
Case	Error F_{\min} (Percent)	Error R_N (Percent)	Error $ \Gamma_0 $ (Percent)	Error $\langle \Gamma_0 \rangle$ (Degree)		
1	3.57	71.86	19.75	4.78		
2	3.34	12.60	8.11	2.36		

Case 1 = source admittances as in [1, table 1] (nine sets).

Case 2 = source admittances as in (14) (seven sets).

The errors are obtained by perturbing the reflection coefficients with a vector of magnitude 0.02 and a phase varying in steps of 1° .

The software can also perform an error sensitivity analysis. In this analysis, the effect of errors is simulated by adding to a source reflection coefficient a vector of small magnitude (entered by the user) and varying phase (0 – 360° with phase steps entered by the user), while keeping the rest of source reflection coefficients constant. The noise parameters are fitted for each possible value of this perturbation vector and compared with their unperturbed values. This process is repeated until all the source reflection coefficients have been perturbed and results in the determination of the perturbed noise parameters that deviate the most with respect to their unperturbed counterparts. The errors between these two sets of noise parameters give an indication of the sensitivity of the noise parameters to errors in the source reflection coefficients. This perturbation analysis is done at each frequency point.

VI. EXECUTION EXAMPLE

The algorithm was tested with the measured data reported in [1, table 1]. In that work, nine measurement sets were taken, but only four of them were used at a time to calculate the noise parameters. This was done for a number of combinations of four data sets. In each case, the noise parameters were obtained and an error function was calculated to assess the agreement between these noise parameters and the nine measurement sets. Proper noise parameter values are assumed when the value of such error function is small. By using sets 3, 4, 7, and 9, the following results were reported [1]:

$$\begin{aligned} F_{\min} &= 0.4567 \text{ dB} & R_N &= 6.077 \Omega \\ 1/G_0 &= 38.2 \Omega & 1/B_0 &= 40.65 \Omega \end{aligned}$$

The same results were found with our software by considering only the above sets in the determination of the noise parameters. The relative RMS error among the measurement sets, defined as

$$\text{ERR} = \frac{1}{n} \sqrt{\sum_n \left(\frac{F_{\text{meas}} - F_{\text{calc}}}{F_{\text{meas}}} \right)^2} \quad (16)$$

with n (number of sets) = 9, was 8.8×10^{-4} .

Table I compares these results with others obtained with the computation approach presented here. First the

noise parameters considering all the sets are determined. By progressive elimination of the set with highest relative error $((F_{\text{meas}} - F_{\text{calc}})/F_{\text{meas}})$, a combination of four data sets is found whose error is lower than the one reported in [1].

The results in Table I suggest that nonredundant noise parameter determination may give acceptable results provided that all possible combinations (in this reported case $(9_4) = 126$) are checked; however, this might be a slow technique and it is unlikely to perform better than those that minimize the overall error with noniterative techniques.

The sensitivity of the noise parameters to errors in the source reflection coefficients has been compared for the source reflection coefficients shown in (14) and those used in [1, table 1]. The noise figures in the latter were substituted for the errorless values that they would have with the noise parameters shown in the last row of Table I of this paper. Similarly, the theoretical noise figures were calculated for the group of reflection coefficients of (14). A sensitivity analysis was performed in each case following the procedure described in Section V. The magnitude of the perturbation error was 0.02 and its phase was changed in 1° steps. The maximum errors between the perturbed noise parameters and the unperturbed ones are shown in Table II, together with the associated $\cos(\bar{V}_i, \bar{V}_j)$ factors. The group of reflection coefficients of (14) presents a much lower error sensitivity than the one in [1, table 1], even though the latter contains two extra measurement sets. The values of $\cos(\bar{V}_i, \bar{V}_j)$ show a clear correlation between sensitivity and orthogonality of the vectors $\bar{V}_1, \bar{V}_2, \bar{V}_3$, and \bar{V}_4 .

VII. CONCLUSION

Vector calculus concepts simplify the algorithm for noise parameter fitting and provide simple criteria for the selection of source admittances. Noise parameter fitting is made with an algorithm that approximates a vector whose components are the measured noise figures with a linear combination of vectors whose components depend on the source admittances. The coefficients of this linear combination determine the noise parameters through simple equations. Ill conditioning occurs when two or more of the approximating vectors (or linear combinations of them)

tend to be aligned or when the uncertainty in a vector is comparable to its magnitude. Conversely, low sensitivity of the noise parameters to error in the source reflection coefficients is achieved when the components of the approximating vectors do not change rapidly with small perturbations in the source reflection coefficients and when these vectors tend to be orthogonal. These two criteria help select the source admittances. A group of seven source admittances has been chosen following these rules. Its performance is compared with that of a group of nine admittances used in a previous work [1]. Computer-simulated errors in the reflection coefficients of the first group generate errors in the noise parameters that are much smaller than those obtained when the second group is used.

APPENDIX I

FORMULAS FOR c_i FROM CHOLESKY'S METHOD

Intermediate Variables

$$\begin{aligned}
 m_{11} &= |V_1| \\
 m_{21} &= \frac{\langle V_2 V_1 \rangle}{m_{11}} & m_{31} &= \frac{\langle V_3 V_1 \rangle}{m_{11}} & m_{41} &= \frac{\langle V_4 V_1 \rangle}{m_{11}} \\
 m_{22} &= \sqrt{|V_2|^2 - m_{21}^2} & m_{32} &= \frac{1}{m_{22}} [\langle V_3 V_2 \rangle - m_{31} m_{21}] \\
 m_{42} &= \frac{1}{m_{22}} [\langle V_4 V_2 \rangle - m_{41} m_{21}] \\
 m_{33} &= \sqrt{|V_3|^2 - m_{31}^2 - m_{32}^2} \\
 m_{43} &= \frac{1}{m_{33}} [\langle V_4 V_3 \rangle - m_{41} m_{31} - m_{42} m_{32}] \\
 m_{44} &= \sqrt{|V_4|^2 - m_{41}^2 - m_{42}^2 - m_{43}^2} \\
 y_1 &= \frac{\langle \bar{V}_1 \bar{F}_M \rangle}{m_{11}} & y_2 &= \frac{1}{m_{22}} [\langle \bar{V}_2 \bar{F}_M \rangle - m_{21} y_1] \\
 y_3 &= \frac{1}{m_{33}} [\langle \bar{V}_3 \bar{F}_M \rangle - m_{31} y_1 - m_{32} y_2] \\
 y_4 &= \frac{1}{m_{44}} [\langle \bar{V}_4 \bar{F}_M \rangle - m_{41} y_1 - m_{42} y_2 - m_{43} y_3].
 \end{aligned}$$

Coefficients c_i

$$\begin{aligned}
 c_4 &= \frac{y_4}{m_{44}} & c_3 &= \frac{1}{m_{33}} [y_3 - m_{43} c_4] \\
 c_2 &= \frac{1}{m_{22}} [y_2 - m_{32} c_3 - m_{42} c_4] \\
 c_1 &= \frac{1}{m_{11}} [y_1 - m_{21} c_2 - m_{31} c_3 - m_{41} c_4].
 \end{aligned}$$

APPENDIX II

ADMITTANCE LOCI THAT PRODUCE ILL CONDITIONING

$$\left(G_s - \frac{1}{2\alpha_1} \right)^2 + \beta_s^2 = \frac{1}{4\alpha_1^2}$$

$$G_s - \alpha_2 B_s = 0$$

$$G_s = \alpha_3$$

$$G_s^2 + \left(B_s - \frac{\alpha_4}{2} \right)^2 = \frac{\alpha_4^2}{4}$$

$$G_s^2 + B_s^2 = \alpha_5$$

$$B_s = \alpha_6$$

$$\left(G_s - \frac{1}{2\alpha_7} \right)^2 + \left(B_s + \frac{\beta_7}{2\alpha_7} \right)^2 = \frac{1}{4\alpha_7^2} (1 + \beta_7^2)$$

$$\left(G_s - \frac{1}{2\alpha_8} \right)^2 + B_s^2 = \frac{1}{4\alpha_8} \left(\frac{1}{\alpha_8} - 4\beta_8 \right)$$

$$G_s - \alpha_9 B_s = \beta_9$$

$$G_s^2 + \left(B_s - \frac{\alpha_{10}}{2} \right)^2 = \frac{\alpha_{10}^2}{4} + \beta_{10}$$

$$\left(G_s - \frac{1}{2\alpha_{11}} \right)^2 + \left(B_s + \frac{\beta_{11}}{2\alpha_{11}} \right)^2 = \frac{1}{4\alpha_{11}} (1 + \beta_{11})^2 - \frac{\gamma_{11}}{\alpha_{11}}.$$

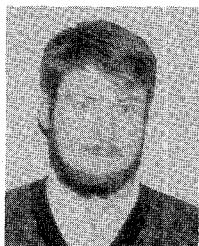
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